

Correspondence

On a Sub-Program to determine the Rayleigh Quotient by Conjugate Gradient Method

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Introduction

This paper is represented for a sub-program in order to determine the eigenvalue with respect to the continuous elastic system by applying the conjugate gradient method to Rayleigh quotient. This numerically computational method has been known as iterative method, and frequently used in order to compute the algebraic equation with several variables or large dimensional algebraic equation for elastic-plastic problems by Yamada [1]. The author also reported for eigenvalue problems of the continuous system in which this method were applied to Rayleigh quotient [2].

The advantages which this method is used are expressed in Ref. [3], therefore they are abbreviated in this paper. In applying this method Rayleigh quotient is used in order to numerically compute the eigenvalue instead of directly computing its from other method. The results are illustrated for the typical engineering examples.

1 Analysis System for Eigenvalue

Consider the continuous elastic system subjected to the concentrated axial forces, as shown in Fig. 1. In which ρ , A , EI and L are density per unit length, cross-sectional area, bending rigidity and total length of beam, respectively.

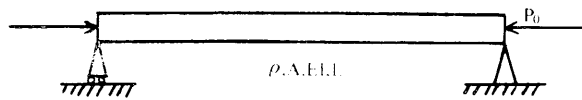


Fig. 1. Simply supported beam subjected to axial forces

The characteristic equations of eigenvalue of dynamic and static problems are generally expressed as,

$$\{[Ke] - P_0[Kg] - \Omega^2[M]\} \bar{q} = 0, \quad (1)$$

or

$$\{[Ke] - \lambda[Kg]\} = 0. \quad (2)$$

In which $[Ke]$, $[Kg]$ and $[M]$ are total stiffness, stability and mass matrices, respectively, Ω eigen circular frequency when the beam is subjected to the concentrated axial loads,

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λ critical value of buckling. Note that Eqs. (1) and (2) are formulated by the standard finite element method. The solution of the above equations can be obtained by computing the determinant as follow,

$$|[Ke] - P_0[Kg] - \Omega^2[M]| = 0, \quad (3)$$

or

$$|[Ke] - \lambda[Kg]| = 0. \quad (4)$$

Note that eigen circular frequency of free vibration can be directly obtained from Eq.(3) in $P_0=0$.

2 Algorithm of Conjugate Gradient Method

In this paper, Rayleigh quotient is used in order to obtained the eigenvalue instead of directly computing of Eqs. (3) and (4) stated above. The algorithm is shown as follow. The Rayleigh quotient for dynamic and static problems is expressed as,

$$R_i = \frac{\vec{q}_i^T [M] \vec{q}_i}{\vec{q}_i^T [K] \vec{q}_i}, \quad (5)$$

in which note that matrix $[Kg]$ in the stability problems is corresponding to matrix $[M]$ in the dynamic problems.

The value of this quotient will be a minimum when the displacement vector \vec{q} is that which most closely approximates the actual mode shape. The applying the conjugate gradient method to Eq. (5), it is, in the first, necessary to find the \vec{g}_i of R_i . Differentiating Eq. (5) with respect to \vec{q}_i^T in order to obtain \vec{g}_i yields subsequently,

$$\frac{\partial R_i}{\partial \vec{q}_i^T} = \frac{2}{\vec{q}_i^T [M] \vec{q}_i} ([K] \vec{q}_i - R_i [M] \vec{q}_i),$$

therefore, the direction of gradient is defined by

$$\vec{g}_i = [K] \vec{q}_i - R_i [M] \vec{q}_i. \quad (6)$$

The minimization proceeds step i to step $i+1$ by computing vector \vec{q}_{i+1} in step $i+1$ from vector \vec{q}_i in step i as follow,

$$\vec{q}_{i+1} = \vec{q}_i + \alpha_i \vec{P}_i.$$

where vector \vec{P}_{i+1} is orthogonal with respect to $[K]$, and α_i is determined by minimizing R_i . The substitution of Eq. (7) into Eq. (5) and its differentiation with respect to α_i yields subsequently,

$$\frac{\partial R_i}{\partial \alpha_i} = \frac{1}{\vec{q}_i^T [M] \vec{q}_i} (\vec{P}_i^T [M] \vec{q}_{i+1})(\vec{q}_i^T [M] \vec{q}_{i+1}) - (\vec{q}_{i+1}^T [K] \vec{q}_{i+1})(\vec{P}_i^T [M] \vec{q}_{i+1}). \quad (7)$$

The minimum value of R will occur where this partial derivative in zero. The extreme value will be obtained by solving the following equation,

$$(\vec{P}_i^T [K] \vec{q}_{i+1})(\vec{q}_i^T [M] \vec{q}_{i+1}) - (\vec{q}_{i+1}^T [K] \vec{q}_{i+1})(\vec{P}_i^T [M] \vec{q}_{i+1}) = 0. \quad (8)$$

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Substituting Eq. (7) into Eq. (8), the quadratic equation with respect to α_i can be subsequently obtained as,

$$a\alpha_i^2 + b\alpha_i + c = 0,$$

where the coefficients a , b and c are of the forms as,

$$a = (\vec{P}_i^T [K] \vec{P}_i)(\vec{q}_i^T [K] \vec{P}_i) - (\vec{q}_i^T [K] \vec{P}_i)(\vec{P}_i^T [M] \vec{P}_i),$$

$$b = (\vec{P}_i^T [K] \vec{P}_i)(\vec{q}_i^T [M] \vec{q}_i) - (\vec{q}_i^T [K] \vec{q}_i)(\vec{q}_i^T [M] \vec{P}_i),$$

$$c = (\vec{q}_i^T [K] \vec{P}_i)(\vec{q}_i^T [M] \vec{q}_i) - (\vec{q}_i^T [K] \vec{q}_i)(\vec{P}_i^T [M] \vec{q}_i).$$

The two roots with respect to α_i are computed from

$$\alpha_i = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a} ; (i = 1, 2).$$

One of the two roots can be selected to minimizing quotient R obtained by substituting these into Eqs. (5) and (7), as shown in Fig. 2.

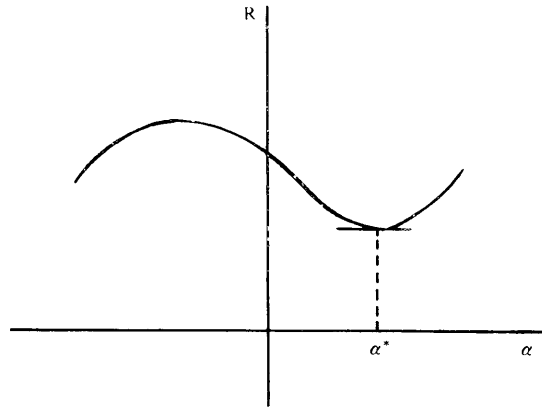


Fig. 2. Representative variation of Rayleigh quotient a line

Vector \vec{P}_{i+1} in step $i+1$ is determined from the gradient \vec{g}_{i+1} and vector \vec{P}_i obtained from the previous step by using the following formula,

$$\vec{P}_{i+1} = \vec{g}_{i+1} + \beta_i \vec{P}_i \quad (9)$$

where β_i is a constant coefficient and is found by requiring that \vec{P}_i and \vec{P}_{i+1} are orthogonal with respect to $[K]$. Therefore the value of β_i is given by,

$$\beta_i = -\frac{\vec{P}_i^T [K] \vec{g}_{i+1}}{\vec{P}_i^T [K] \vec{P}_i} \quad (10)$$

The algorithm can be summarized as,

1. Initialization ($i=0$)

\vec{q}_0 = given arbitrary constant,

$$\vec{g}_0 = R(\vec{q}_0),$$

$$\vec{P}_0 = \vec{g}_0.$$

2. Next step iteration

(1) compute α_i ,

(2) compute $\vec{q}_{i+1} = \vec{q}_i + \alpha_i \vec{P}_i$,

(3) compute $R_{i+1} = \frac{\vec{q}_{i+1}^T [K] \vec{q}_{i+1}}{\vec{q}_{i+1}^T [M] \vec{q}_{i+1}}$

(4) compute $\vec{g}_{i+1} = [K] \vec{q}_{i+1} - R_{i+1} [M] \vec{q}_{i+1}$

(5) compute $\beta = -\frac{\vec{P}_i^T [K] \vec{g}_{i+1}}{\vec{P}_i^T [K] \vec{P}_i}$

(6) compute $\vec{P}_{i+1} = \vec{g}_{i+1} + \beta \vec{P}_i$

The iteration are carried out to obtained the convergent value of R . The then convergent value can be given as follow,

$$|R_{i+1} - R_i| < e,$$

where e is the condition of coverage.

3 Numerical Example and Conclusion

Consider the simply supported uniform beam subjected to the concentrated axial forces at both ends, as shown in Fig. 1. In which mass, stiffness and stability matrices can be obtained by means of applying the standard finite element method, respectively. These are denoted as follow,

$$[m] = \frac{\rho A l}{420} \begin{bmatrix} 156 & 22l & 54 & -13l \\ 22l & 4l^2 & 13l & -3l^2 \\ 54 & 13l & 156 & -22l \\ -13l & -3l^2 & -22l & 4l^2 \end{bmatrix},$$

$$[k_e] = \frac{EI}{l^3} \begin{bmatrix} 12 & 6l & -12 & 6l \\ 6l & 4l^2 & -6l & 2l^2 \\ -12 & -6l & 12 & -6l \\ 6l & 2l^2 & -6l & 4l^2 \end{bmatrix},$$

$$[k_g] = \frac{P_0}{l} \begin{bmatrix} 6/5 & l/10 & -6/5 & l/10 \\ l/10 & 2l^2/15 & -l/10 & -l^2/30 \\ -6/5 & -l/10 & 6/5 & -l/10 \\ l/10 & -l^2/30 & -l/10 & 2l^2/15 \end{bmatrix}.$$

Each of the above element matrices must be assembled for the given entire structural system when these are employed to practical problems. R value corresponding to

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eigenvalue, ω^2 , of transverse free vibration and critical buckling value, P_{cr} , can be expressed as follow,

$$R = \omega^2 \frac{\rho AL^4}{EI} = \frac{\tilde{q}_i^T [Ke] \tilde{q}_i}{\tilde{q}_i^T [M] \tilde{q}_i} = \pi^4,$$

$$R = P_{cr} \frac{L^2}{EI} = \frac{\tilde{q}_i^T [Ke] \tilde{q}_i}{\tilde{q}_i^T [Kg] \tilde{q}_i} = \pi^2.$$

The results are shown for two kinds of the finite element in Table 1. The each of (2) and (4) in Table 1 expressed on the divided number of finite elements. The results are expressed that the four divided finite elements are sufficiently satisfied when the computation is carried out for the practical problems.

Table 1 A Comparison of eigenvalues

	$\sqrt{\omega^2 \frac{\rho AL^4}{EI}}$	error %	$P_{cr} \frac{L^2}{EI}$	error %
Theory	9.87		9.87	
Present solution	9.91(2)	0.4	9.94(2)	0.7
	9.88(4)	0.1	9.89(4)	0.2

The sub-program is shown in Appendix. Note that $EPS=10^{-5}$ for the condition of convergence and total length $L=450$ [mm] of the beam are given in order to carry out this computation.

4 Acknowledgement

The FACOM 230-38 in Kokushikan University Computer Center has been used for the objective computation.

(Recieved 18 January 1982)

References

- 1) Yamada, *et al.*, EPIC-IV, (in Japanese), (1980), BAIFUKAN.
- 2) K. Shimoyamada, *Prescript JSME*, (in Japanese), No. 760-1, (June, 1976), p 69.
- 3) K. Shimoyamada, *Memoris of the Kokushikan University Computer Center*, No.2, (march, 1981), p 15.
- 4) Rubinstein, M.F., *et al.*, *J. Franklin Inst.* Vol. 293, No. 3, (march, 1972), p173.

Appendix

This appendix shows the program for algorithm represented in this paper. When this program is used, we would be required to prepare the subroutine in order to operate, for example, the form of " $\tilde{q}^T [K] \tilde{q}$ " and to clear the vector (VECMAT and CLVEC in this program, respectively)

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SOURCE      LIST
C          *** CONJUGATE GRADIENT METHOD *****
C          AK-CORRESPONDING TO STIFFNESS MATRIX
C          BM-CORRESPONDING TO MASS OR STABILITY MATRIX
C          Q=DISPLACEMENT VECTOR
C          D=JUDGEMENT OF ROOT OF QUADRATIC EQUATION
C          NRED= DIMENSION FOR CALCULATION
C          NELM= DIMENSION OF MATRIX
0001        SUBROUTINE FUMCG(AK,BM,RAL,D,NRED,NELM,Q,EPST)
C          **
0002          DIMENSION AK(NELM,NELM),BM(NELM,NELM),Q(NELM)
0003          DIMENSION P(30),Q(30),D(30),OLDP(30),OLDQ(30),C(30),GG(30)
C          **
C          COMPUTE FOR INITIAL ARGUMENT
0004          CALL CLVEC(Q,NELM)
C          ARBITRARY VALUE GIVEN TO ALL COMPONENT OF DISPLACEMENT Q(I)
0005          DO 20 I=1,NRED
0006            Q(I)=1.0
C          COMPUTE THE EACH FACTOR FOR INITIAL RAYLEIGH QUOTIENT
0007          CALL CLVEC(C,NELM)
0008          CALL VECMAT(C1,Q,Q,AK,C,NRED,NELM)
0009          CALL CLVEC(C,NELM)
0010          CALL VECMAT(C2,Q,Q,BM,C,NRED,NELM)
C          COMPUTE THE INITIAL RAYLEIGH QUOTIENT
0011          HAL=CL/C2
0012          WRITE(6,50) RAL
C          COMPUTE THE GRADIENT VECTOR GG(I)
0013          DO 60 I=1,NRED
0014            S=0.0
0015            DO 70 J=1,NRED
0016              S=S+(AK(I,J)-HAL*BM(I,J))*Q(J)
0017            70 CONTINUE
0018            GG(I)=S
0019          60 CONTINUE
C          *** ORTHOGONAL VECTOR ***** P(I) ****
0020          DO 80 I=1,NRED
0021            P(I)=GG(I)
0022          80 CONTINUE
C          *** NEXT STEP ITERATION *****
0023          K=0
0024          1000 CONTINUE
0025          K=K+1
0026          R=HAL
0027          DO 90 I=1,NRED
0028            OLDQ(I)=Q(I)
0029            OLDP(I)=P(I)
0030          90 CONTINUE
0031          CALL CLVEC(P,NELM)
0032          CALL CLVEC(Q,NELM)
0033          CALL CLVEC(GG,NELM)
C          COMPUTE THE FACTOR OF QUADRATIC EQUATION
0034          CALL VECMAT(D1,OLDP,OLDP,AK,C,NRED,NELM)
0035          CALL CLVEC(C,NELM)
0036          CALL VECMAT(D2,OLDQ,OLDQ,BM,C,NRED,NELM)
0037          CALL CLVEC(C,NELM)
0038          CALL VECMAT(D3,OLDQ,OLDP,AK,C,NRED,NELM)
0039          CALL CLVEC(C,NELM)
0040          CALL VECMAT(D4,OLDP,OLDP,BM,C,NRED,NELM)
0041          AA=(D1+D2)-(D3+D4)
0042          CALL CLVEC(C,NELM)
0043          CALL VECMAT(D1,OLDP,OLDP,AK,C,NRED,NELM)
0044          CALL CLVEC(C,NELM)
0045          CALL VECMAT(D2,OLDQ,OLDQ,BM,C,NRED,NELM)
0046          CALL CLVEC(C,NELM)
0047          CALL VECMAT(D3,OLDQ,OLDQ,AK,C,NRED,NELM)
0048          CALL CLVEC(C,NELM)
0049          CALL VECMAT(D4,OLDP,OLDP,BM,C,NRED,NELM)
0050          CALL CLVEC(C,NELM)
0051          BB=(D1+D2)-(D3+D4)
0052          CALL VECMAT(D1,OLDP,OLDQ,AK,C,NRED,NELM)
0053          CALL CLVEC(C,NELM)
0054          CALL VECMAT(D2,OLDQ,OLDQ,BM,C,NRED,NELM)
0055          CALL CLVEC(C,NELM)
0056          CALL VECMAT(D3,OLDQ,OLDQ,AK,C,NRED,NELM)
0057          CALL CLVEC(C,NELM)
0058          CALL VECMAT(D4,OLDP,OLDQ,BM,C,NRED,NELM)
0059          CC=(D1+D2)-(D3+D4)
0060          CALL QUAD(AA,BB,CC,D,X1,X2)
0061          IF(D.LT.0.0) GO TO 1520
C          *** COMPUTE THE DISPLACEMENT VECTOR QQ QD,*****
0062          DO 100 I=1,NRED
0063            QS(I)=OLDQ(I)*X1+OLDP(I)
0064            QD(I)=OLDQ(I)*X2+OLDP(I)
0065          100 CONTINUE
C          *** COMPUTE THE RAYLEIGH QUOTIENT HAL IN N-STEP
0066          HAL=0.0
0067          CALL CLVEC(C,NELM)
0068          CALL VECMAT(D1,Q,Q,AK,C,NRED,NELM)
0069          CALL CLVEC(C,NELM)
0070          CALL VECMAT(D2,Q,Q,BM,C,NRED,NELM)
0071          CALL CLVEC(C,NELM)
0072          CALL VECMAT(D3,QD,QD,AK,C,NRED,NELM)
0073          CALL CLVEC(C,NELM)
0074          CALL VECMAT(D4,QD,QD,BM,C,NRED,NELM)
0075          HAL1=D1/D2
0076          HAL2=D3/D4
0077          IF(HAL1.LT.HAL2.AND.HAL1.GT.0.0) GO TO 2000
0078          HAL=HAL2
0079          IF(HAL.LT.0.0) GO TO 1500

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SOURCE      LIST
0080          DO 110 I=1,NRED
0081          W(I)=WD(I)
0082          110 CONTINUE
0083          IF (ABS(N-RAL).LT.EPS) GO TO 1510
0084          GO TO 1000
0085          2000 CONTINUE
0086          RAL=RAL1
0087          IF (RAL.LT.C.0) GO TO 1500
0088          DO 120 I=1,NRED
0089          120 W(I)=W(I)
0090          IF (ABS(N-RAL).LT.EPS) GO TO 1510
0091          C      **** COMPUTE THE N-STEP GRADIENT VECTOR *****
0092          1600 CONTINUE
0093          DO 130 I=1,NRED
0094          S=0
0095          DO 140 J=1,NRED
0096          S=S+(X(I,J)-RAL*BM(I,J))*W(J)
0097          140 CONTINUE
0098          GG(I)=S
0099          C      *** COMPUTE THE ORTHOGONAL VECTOR P TO AK
0100          CALL CLVEC(C,NELM)
0101          CALL VECMAT(D1,OLDP,GG,AK,C,NRED,NELM)
0102          CALL CLVEC(C,NELM)
0103          CALL VECMAT(D2,OLDP,OLDP,AK,C,NRED,NELM)
0104          BETA=(D1/D2)
0105          DO 160 I=1,NRED
0106          P(I)=(GG(I)+BETA*OLDP(I))
0107          160 CONTINUE
0108          CALL CLVEC(C,NELM)
0109          CALL CLVEC(OLDP,NELM)
0110          GO TO 1000
0111          1510 WRITE(6,200) K
0112          WRITE(6,180) HAL1,HAL2,RAL
0113          RETURN
0114          1500 WRITE(6,170)
0115          RETURN
0116          1520 WRITE(6,190)
0117          RETURN
0118          50 FORMAT(/5X,4H RAL=,E15.7)
0119          170 FORMAT(/5X,30X,'0th= OBTAINED NEGATIVE RALEIGH QUOTIENT= ')
0120          180 FORMAT(1H ,5X,5H RAL1=,E15.7,10X,5H RAL2=,E15.7,10X,4H RAL=,E15.7)
0121          190 FORMAT(/5X,' *OBTAINED NEGATIVE ROOT OF QUADRATIC EQUATION * ')
0122          200 FORMAT(1H ,15,1X,14H STEP ITERATION)
0123          END

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